

TEMA: 0621 ATP-RTC - Aerodynamics - Chap.3

COD_PREG: PREGUNTA: RPTA:
PREG20097876 (8345) What effect does an increase in airspeed have on a coordinated turn while maintaining a constant angle of bank and altitude? C
OPCION A: The rate of turn will decrease resulting in a decreased load factor
OPCION B: The rate of turn will increase resulting in an increased load factor
OPCION C: The rate of turn will decrease resulting in no changes in load factor
OPCION D:

PREG20097877 (8346) What is the effect on total drag of an aircraft if the airspeed decreases in level flight below that speed for maximum L/D? A
OPCION A: Drag increases because of increased induced drag.
OPCION B: Drag increases because of increased parasite drag
OPCION C: Drag decreases because of lower induced drag
OPCION D:

PREG20097878 (8347) What is load factor? C
OPCION A: Lift multiplied by the total weight
OPCION B: Lift subtracted from the total weight
OPCION C: Lift divided by the total weight
OPCION D:

PREG20097879 (8348) What affects indicated stall speed? A
OPCION A: Weight, load factor, and power
OPCION B: Load factor, angle of attack, and power
OPCION C: Angle of attack, weight, and air density
OPCION D:

PREG20097880 (8349) If no corrective action is taken by the pilot as angle of bank is increased, how is the vertical component of lift and sink rate affected? C
OPCION A: Lift increases and the sink rate increases
OPCION B: Lift decreases and the sink rate decreases
OPCION C: Lift decreases and the sink rate increases
OPCION D:

PREG20097881 (8350) Why must the angle of attack be increased during a turn to maintain altitude? A
OPCION A: Compensate for loss of vertical component of lift
OPCION B: Increase the horizontal component of lift equal to the vertical component
OPCION C: Compensate for increase in drag

OPCION D:

PREG20097882 (8351) How can the pilot increase the rate of turn and decrease the radius at the same time? B

OPCION A: Steepen the bank and increase airspeed

OPCION B: Steepen the bank and decrease airspeed

OPCION C: Shallow the bank and increase airspeed

OPCION D:

PREG20097883 (8352) What is the relationship of the rate of turn with the radius of turn with a constant angle of bank but increasing airspeed? A

OPCION A: Rate will decrease and radius will increase

OPCION B: Rate will increase and radius will decrease

OPCION C: Rate and radius will increase

OPCION D:

PREG20097884 (8353) Upon which factor does wing loading during a level coordinated turn in smooth air depend? B

OPCION A: Rate of turn

OPCION B: Angle of bank

OPCION C: True airspeed

OPCION D:

PREG20097885 (8354) If an aircraft with a gross weight of 2,000 pounds were subjected to a total load of 6,000 pounds in flight, the load factor would be B

OPCION A: 2 Gs.

OPCION B: 3 Gs

OPCION C: 9 Gs.

OPCION D:

PREG20097886 (8355) What is the ratio between the total air load imposed on the rotor disc and the gross weight of a helicopter in flight? B

OPCION A: Power loading.

OPCION B: Load factor.

OPCION C: Aspect ratio.

OPCION D:

PREG20097887 (8365) Identify the type stability if the aircraft attitude remains in the new position after the controls have been neutralized C

OPCION A: Negative longitudinal static stability

OPCION B: Neutral longitudinal dynamic stability

OPCION C: Neutral longitudinal static stability

OPCION D:

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| PREG20097888 (8368) | What is the reason for variations in geometric pitch along a propeller or rotor blade? | A |
| OPCION A: | It permits a relatively constant angle of attack along its length when in cruising flight. | |
| OPCION B: | It prevents the portion of the blade near the hub or root from stalling during cruising flight. | |
| OPCION C: | It permits a relatively constant angle of incidence along its length when in cruising flight. | |
| OPCION D: | | |

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| PREG20097889 (8372) | Identify the type stability if the aircraft attitude tends to move farther from its original position after the controls have been neutralized | A |
| OPCION A: | Negative static stability | |
| OPCION B: | Positive static stability | |
| OPCION C: | Negative dynamic stability | |
| OPCION D: | | |

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| PREG20097890 (8373) | Identify the type stability if the aircraft attitude tends to return to its original position after the controls have been neutralized | B |
| OPCION A: | Positive dynamic stability | |
| OPCION B: | Positive static stability | |
| OPCION C: | Neutral dynamic stability | |
| OPCION D: | | |

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| PREG20097891 (8375) | What flight condition should be expected when an aircraft leaves ground effect? | A |
| OPCION A: | An increase in induced drag requiring a higher angle of attack | |
| OPCION B: | A decrease in parasite drag permitting a lower angle of attack | |
| OPCION C: | An increase in dynamic stability | |
| OPCION D: | | |

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| PREG20097892 (8376) | What characteristic should exist if an airplane is loaded to the rear of its CG range? | C |
| OPCION A: | Sluggish in aileron control | |
| OPCION B: | Sluggish in rudder control | |
| OPCION C: | Unstable about the lateral axis | |
| OPCION D: | | |

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| PREG20097893 (8377) | What will be the ratio between airspeed and lift if the angle attack and other factors remain constant and airspeed is doubled? Lift will be | C |
| OPCION A: | the same | |
| OPCION B: | two times greater | |
| OPCION C: | four times greater | |
| OPCION D: | | |

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| PREG20097894 (8378) | What true airspeed and angle of attack should be used to generate the same of lift as altitude is increased? | B |
| OPCION A: | The same true airspeed and angle of attack | |
| OPCION B: | A higher true airspeed for any given angle of attack | |
| OPCION C: | A lower true airspeed and higher angle of attack | |
| OPCION D: | | |

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| PREG20097895 (8396) | For a given angle of bank, the load factor imposed on both the aircraft and pilot in a coordinated constant-altitude turn | C |
| OPCION A: | is directly related to the airplane's gross weight | |
| OPCION B: | varies with the rate of turn | |
| OPCION C: | is constant | |
| OPCION D: | | |

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| PREG20097896 (8397) | What is the relationship between induced and parasite drag when the gross weight is increased? | B |
| OPCION A: | Parasite drag increases more than induced drag | |
| OPCION B: | Induced drag increases more than parasite drag. | |
| OPCION C: | Both parasite and induced drag are equally increased | |
| OPCION D: | | |

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| PREG20097897 (8402) | How should a pilot execute a pinnacle-type approach to a rooftop heliport in conditions of high wind and turbulence? | A |
| OPCION A: | Steeper-than-normal approach, maintaining the desired angle of descent with collective. | |
| OPCION B: | Normal approach, maintaining a slower-than-normal rate of descent with cyclic. | |
| OPCION C: | Shallow approach, maintaining a constant line of descent with cyclic. | |
| OPCION D: | | |

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| PREG20097898 (8403) | How should a quick stop be initiated? | B |
| OPCION A: | Raise collective pitch. | |
| OPCION B: | Apply aft pitch. | |
| OPCION C: | Decrease RPM while raising collective pitch. | |
| OPCION D: | | |

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| PREG20097899 (8404) | How does Vne speed vary with altitude? | C |
| OPCION A: | Varies directly with altitude. | |
| OPCION B: | Remains the same at all altitudes. | |
| OPCION C: | Varies inversely with altitude. | |
| OPCION D: | | |

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| PREG20097900 (8405) | What limits the high airspeed potential of a helicopter? | B |
| OPCION A: | Harmonic resonance. | |
| OPCION B: | Retreating blade stall. | |
| OPCION C: | Rotor RPM limitations. | |
| OPCION D: | | |

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| PREG20097901 (8406) | What corrective action can a pilot take to recover from settling with power? | C |
| OPCION A: | Increase forward speed and raise collective pitch. | |
| OPCION B: | Decrease forward speed and partially raise collective pitch. | |
| OPCION C: | Increase forward speed and partially lower collective pitch. | |
| OPCION D: | | |

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| PREG20097902 (8408) | The lift differential that exists between the advancing main rotor blade and the retreating main rotor blade is known as | B |
| OPCION A: | Coriolis effect. | |
| OPCION B: | dissymmetry of lift. | |
| OPCION C: | translating tendency. | |
| OPCION D: | | |

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| PREG20097903 (8409) | During a hover, a helicopter tends to drift in the direction of a tail rotor thrust. What is the movement called? | A |
| OPCION A: | Translating tendency. | |
| OPCION B: | Transverse flow effect. | |
| OPCION C: | Gyroscopic precession. | |
| OPCION D: | | |

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| PREG20097904 (8410) | What is the purpose of the lead-lag (drag) hinge in a three-bladed, fully articulated helicopter rotor system? | B |
| OPCION A: | Offset lateral instability during autorotation. | |
| OPCION B: | Compensate for Coriolis effect. | |
| OPCION C: | Provide geometric balance. | |
| OPCION D: | | |

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| PREG20097905 (8411) | During an autorotation (collective pitch full down), what is an increase in rotor RPM associated with? | A |
| OPCION A: | An increase in airflow through the rotor system. | |
| OPCION B: | A decrease in airflow through the rotor system. | |
| OPCION C: | A decrease in airspeed. | |
| OPCION D: | | |

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| PREG20097906 (8412) | What corrective action can a pilot take to prevent a retreating blade stall at its onset? | A |
| OPCION A: | Reduce collective pitch and increase rotor RPM. | |

OPCION B: Increase collective pitch and increase rotor RPM.

OPCION C: Reduce collective pitch and decrease rotor RPM.

OPCION D:

PREG20097907 (8413) Which is a major warning of approaching retreating blade stall? C

OPCION A: High frequency vibration.

OPCION B: Tendency to roll opposite the stalled side of the rotor.

OPCION C: Pitchup of the nose.

OPCION D:

PREG20097908 (8417) How does high density altitude affect helicopter performance? B

OPCION A: Engine and rotor efficiency are increased.

OPCION B: Engine and rotor efficiency are reduced.

OPCION C: Engine efficiency is reduced, but rotor efficiency is increased.

OPCION D:

PREG20097909 (8418) How is the helicopter climb performance most adversely affected? A

OPCION A: Higher-than-standard temperature and high relative humidity.

OPCION B: Lower-than-standard temperature and high relative humidity.

OPCION C: Higher-than-standard temperature and low relative humidity.

OPCION D:

PREG20097910 (8420) What causes Coriolis effect? C

OPCION A: Differential thrust of rotor blades.

OPCION B: Changing angle of attack of blades during rotation.

OPCION C: Shift in center of mass of flapping blade.

OPCION D:

PREG20097911 (8421) Why are the rotor blades more efficient when operating in ground effect? A

OPCION A: Induced drag is reduced.

OPCION B: Induced angle of attack is increased.

OPCION C: Downwash velocity is accelerated.

OPCION D:

PREG20097912 (8422) What result does a level turn have on the total lift force and load factor? C

OPCION A: Lift force remains constant and the load factor increases.

OPCION B: Lift force increases and the load factor decreases.

OPCION C: Both total lift force and load factor increase.

OPCION D:

PREG20097913 (8423) What causes a helicopter to turn? B

OPCION A: Centrifugal force.

OPCION B: Horizontal component of lift.

OPCION C: Greater angle of attack of rotor blades on upward side of the rotor disc.

OPCION D:

PREG20097914 (8424) What is the primary purpose of the tail rotor system? C

OPCION A: Maintain heading during forward flight.

OPCION B: Act as a rudder to assist in coordinated turns.

OPCION C: Counteract the torque effect of the main rotor.

OPCION D:

PREG20097915 (8425) Under what condition would it be necessary to cause the tail rotor to direct thrust to the left on an American-made helicopter? B

OPCION A: To maintain heading with a left crosswind.

OPCION B: To counteract the drag of the transmission during autorotation.

OPCION C: To execute hovering turns to the right.

OPCION D:

PREG20097916 (9318) Which statement describes the term "VTOSS"? B

OPCION A: The takeoff safety speed in a turbine-engine powered transport category airplane.

OPCION B: The takeoff safety speed in a Category A helicopter.

OPCION C: The takeoff stall speed in the takeoff configuration in a turbo-propeller powered airplane.

OPCION D:
